Physics 200 (Stapleton)
Circular Motion, Gravity, Kepler
Conceptual Practice, etc.

Name: Answers

Provided Formulas: $a_{centripetal} = v^2/r$

 $F_{centripetal} = ma_{centripetal}$ $G = 6.67x10^{-11}Nm^2/kg^2$

$$F_{gravity} = G\left(\frac{M_1 m_1}{r^2}\right)$$

$$\frac{T_A^2}{T_B^2} = \frac{r_A^3}{r_B^3}$$

$$rac{T_A^2}{T_B^2} = rac{r_A^3}{r_B^3}$$

A car is traveling at a constant speed on a loop-the-loop. Which of the following always represents the net force acting on the car while it is traveling in the circle?

- a. The centripetal force
- b. The weight of the car.

 True or

 Gides
- d. The weight plus the normal force
 - e. The weight minus the centripetal force

never true
If you move twice as far from the Earth's center, your weight...

- a. Doubles
- b. Quadruples c. Is divided by 2
- d. Is divided by 4

3. Planet A is twice as far away from the Sun as Planet B. This means planet A's orbital period takes _____ as $\frac{T_{3}^{2}}{T_{3}^{2}} = \frac{r_{3}^{2}}{r_{3}^{2}} = \frac{T_{4}^{2}}{r_{3}^{2}} = \frac{(2)^{\frac{3}{4}}}{r_{3}^{2}} = \frac{8}{1} \sqrt{\frac{r_{4}^{2}}{r_{3}^{2}}} = \sqrt{8} = \frac{T_{4}}{T_{3}}$ many Earth years. a. 4x b. 8x (c. $\sqrt{8x}$)d. 16x

What is the difference between a satellite's altitude and its orbital radius?

Distance to Earth's surface

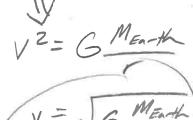
Pistance to Earth's Center 5. A tetherball is rapidly orbiting a central pole. No outside force is being applied to the system. What is pulling the ball away from the pole?

Inertia

Starting with the Universal Law of Gravitation and centripetal force, derive a formula for the speed of a 6. stable orbit around Earth. Show your steps.

MV2 = 6 m MEnth => MV= 6 m MEnth

7. Provide a formula for g on a planet of Mass M and radius r.



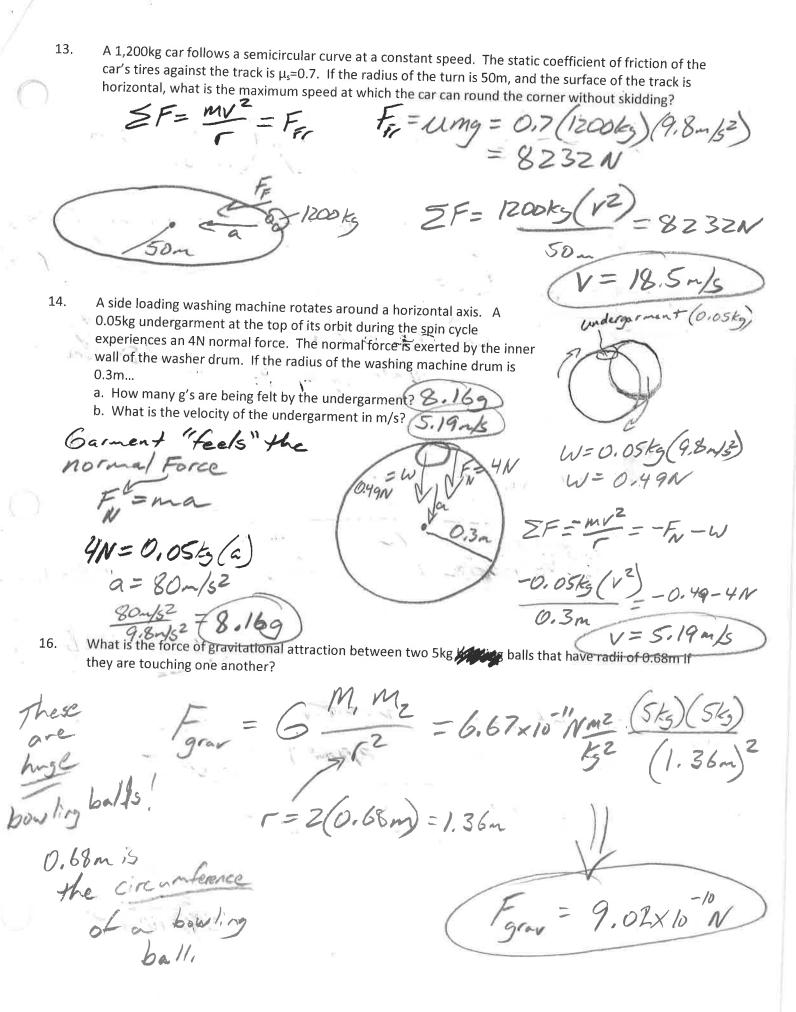
Ta=TRV8

A geostationary satellite orbits the Earth in a circular orbit while remaining over one fixed point on the 8. equator. Can such a satellite orbit at a variety of altitudes or at just one altitude? Use one of Kepler's laws to explain your answer. Just one altitude. TA2 = TA For objects orbiting

TB2 TA3 one planet, the

ratio of period to radius is

fixed. So, for a period of 1 day, Keplers 3rd Law there is only one possible radius. What does Kepler's 1st law tell us about planetary orbits? Orbits are elliptical, with the sun at one focus of the ellipse On the diagram to the right, label the areas where the planet is speeding up 10. speeding up I and slowing down. 5/aving [] Explain why the planet is speeding up and slowing down in those areas of the diagram. 11 In the shaded areas, gravity is pulling in the same direction as the velocity. In the white area gravity opposes velocity/ In 2017, the Earth was at Perihelion (closest to the sun) on January 4th. We 12. will be at aphelion on July 3rd. In the Northern Hemisphere, are our winters longer than our summers, or are our Summers longer than our winters? Sun Which of Kepler's Laws gives us the answer to this question? Explain your reasoning. Summers are longer, Keplos 2nd Law (Equal Areas) says we move faster when were closer to the sun, some spend more time here Which is when



16. Calculate g on Pluto, which has a mass of 1.3x10²²kg and a radius of 1,187km.

$$9 = 6 \frac{M_{\text{plulo}}}{r^2} = 6.67 \times 10^{-11} N_{\frac{m^2}{52}} \left(\frac{1.3 \times 10^{22} k_{5}}{1.187 \times 10^{6} \text{m}^{2}} \right)$$

9=6.15×10 m/s2 = 0.615m/s2

- 17. The Earth's mass is 5.972x10²⁴kg and its average radius is 6.121x10⁶m. Use these numbers to calculate the velocity of a satellite in a stable, circular orbit at a constant altitude of 28,000km above Earth's surface.
 - a. What is the satellite's orbital radius? 3.49 1×10 m
 - b. What is the satellite's velocity 3,402 m/s 9,171×10 m

$$V = \sqrt{\frac{6m_{Earth}}{r}}$$

$$V = \sqrt{\frac{6.67 \times 10^{-11} \times 10^{2}}{5.972 \times 10^{24} \times 3}} \sqrt{\frac{5.972 \times 10^{24} \times 3}{3.44 \times 10^{7} \text{m}}}$$

- 18. a. What is the orbital period of the satellite in the previous question?
 - b. Use your previous answers to find the orbital radius of a geostationary satellite.

$$r_{3}^{3} = 7.54 \times 10^{22}$$